

9.6 Wingtip vortices

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How do airplanes fly?

... is a far more subtle question than you may realize.

Section 9.6 is a neat example but the physical explanation is a bit sketchy. One thing that is not clearly stated is the *Kutta-Zhukovsky theorem*.*

Lift = airspeed · circulation · air density · wingspan

For a good general reference on all things flight-related, see

<http://www.av8n.com/how/>

by John S. Denker. The drawings there are fantastic, much better than your text.

*Zhukovsky is a Russian name, you may see it transliterated many ways, including Joukowski!

Problem a: Is the circulation $\oint_C \vec{v} \cdot d\vec{r}$ positive or negative in fig 9.10 for contour C?

Think about $\vec{v} \cdot d\vec{r}$ as you go around C. On the top, the airflow is both faster and opposite the sense of C, so the top makes a large negative contribution. The bottom of the flow is slow, but aligned with the sense of C, this gives a small positive contribution.

Conclusion: the circulation is *negative*:

$$\oint_C \vec{v} \cdot d\vec{r} < 0. \quad (1)$$

Problem b: See figure 9.11. Just apply Stoke's theorem to the circulation:

$$\oint_C \vec{v} \cdot d\vec{r} = \int_S (\nabla \times \vec{v}) \cdot d\vec{S} \quad (2)$$

For a surface S enclosing the wingtip. The vorticity is just

$$\vec{\omega} = \nabla \times \vec{v}, \quad (3)$$

so Stoke's theorem becomes

$$\oint_C \vec{v} \cdot d\vec{r} = \int_S \vec{\omega} \cdot d\vec{S}. \quad (4)$$

which we sought to show. The circulation around a contour is the sum of the vorticity through a surface bounded by that contour.

The vorticity is $\nabla \times \vec{\mathbf{v}}$. Remember that ∇ involves spatial derivatives. Where will the flow be changing most rapidly in space? Near the wingtips. Wings produce lift when the circulation along $C \neq 0$. Wingtip vortices are associated with lift.

Problem c: does the wingtip vortex in fig 9.11 rotate clockwise (A) or counterclockwise (B)?

Look carefully at the figure. Note $\hat{\mathbf{n}}$ pointing out from S . We know that

$$\oint_C \vec{\mathbf{v}} \cdot d\vec{\mathbf{r}} < 0.$$

This implies that

$$\int_S \vec{\omega} \cdot d\vec{\mathbf{S}} = \int_S (\nabla \times \vec{\mathbf{v}}) \cdot d\vec{\mathbf{S}} < 0. \quad (5)$$

In terms of $\hat{\mathbf{n}}$:

$$\int_S (\nabla \times \vec{\mathbf{v}}) \cdot \hat{\mathbf{n}} dS < 0, \quad (6)$$

So, for the vortex near the tip where the contribution to the integral is strongest,

$$(\nabla \times \vec{\mathbf{v}}) \cdot \hat{\mathbf{n}} = \vec{\omega} \cdot \hat{\mathbf{n}} < 0, \quad (7)$$

which implies the vorticity vector $\vec{\omega}$ points *opposite* to $\hat{\mathbf{n}}$. By the right-hand-rule, this means the vortex is clockwise in the picture, choice **A**.

Now does this make physical sense?

The opposite wing will have the opposite sense for everything, so a counterclockwise vortex. See the figure at:

<http://www.av8n.com/how/htm/airfoils.html#fig-trailing>

The vortices produce descending air behind the plane. The wings exert forces downward on this air, changing its momentum to have a downward component. By Newton's Third law, this produces an upward force on the wings: the lift. When Newton's laws are satisfied, we feel pretty happy.

I heartily recommend the "See how it flies". The author is both a pilot and a physicist.